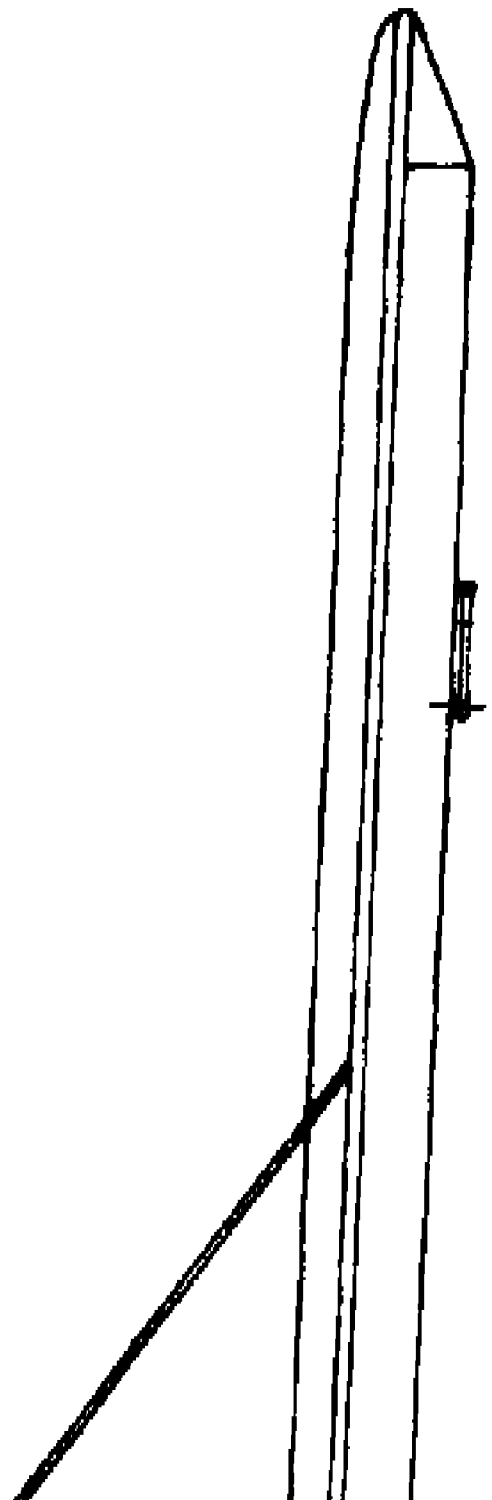


FRONT VIEW

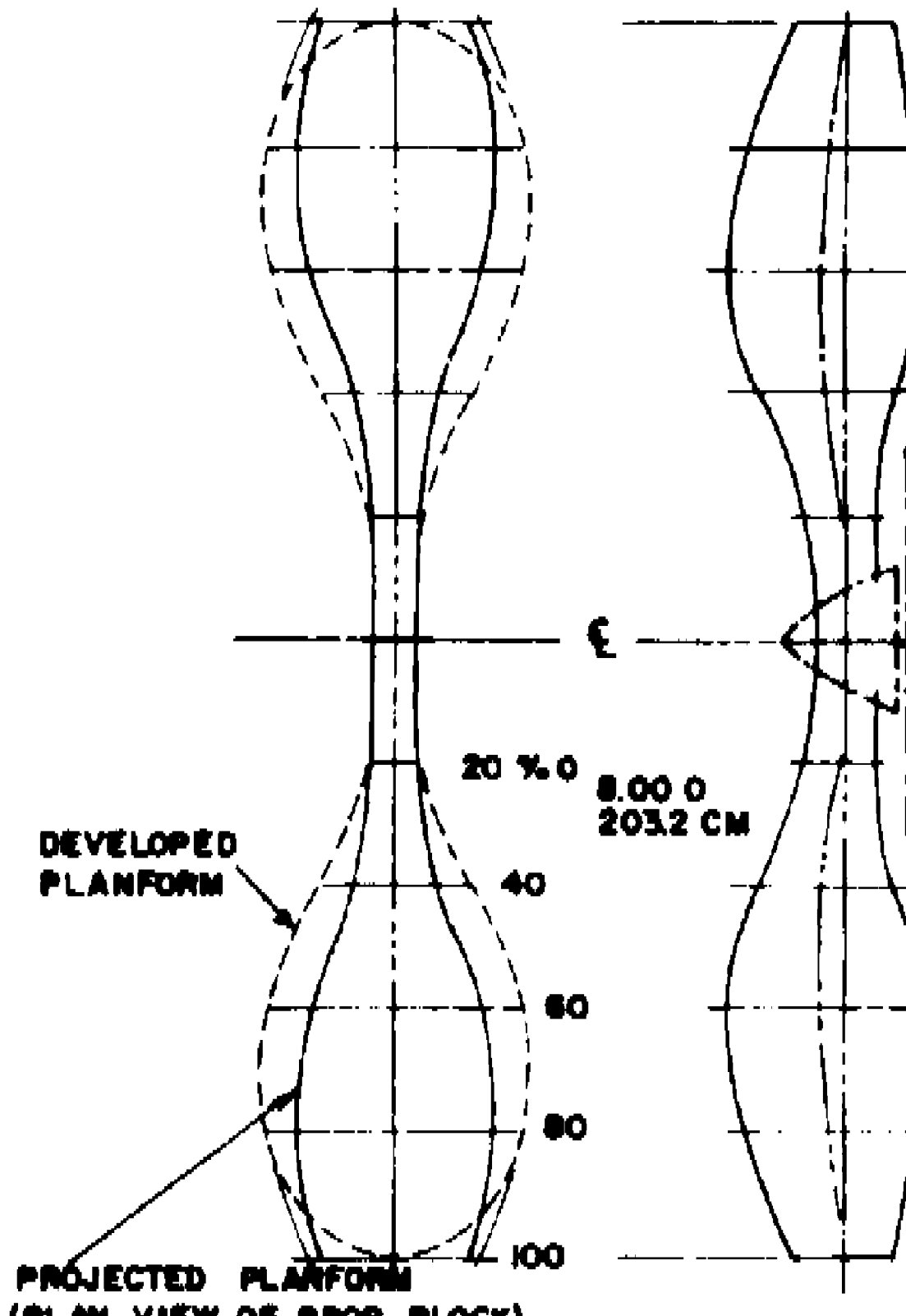
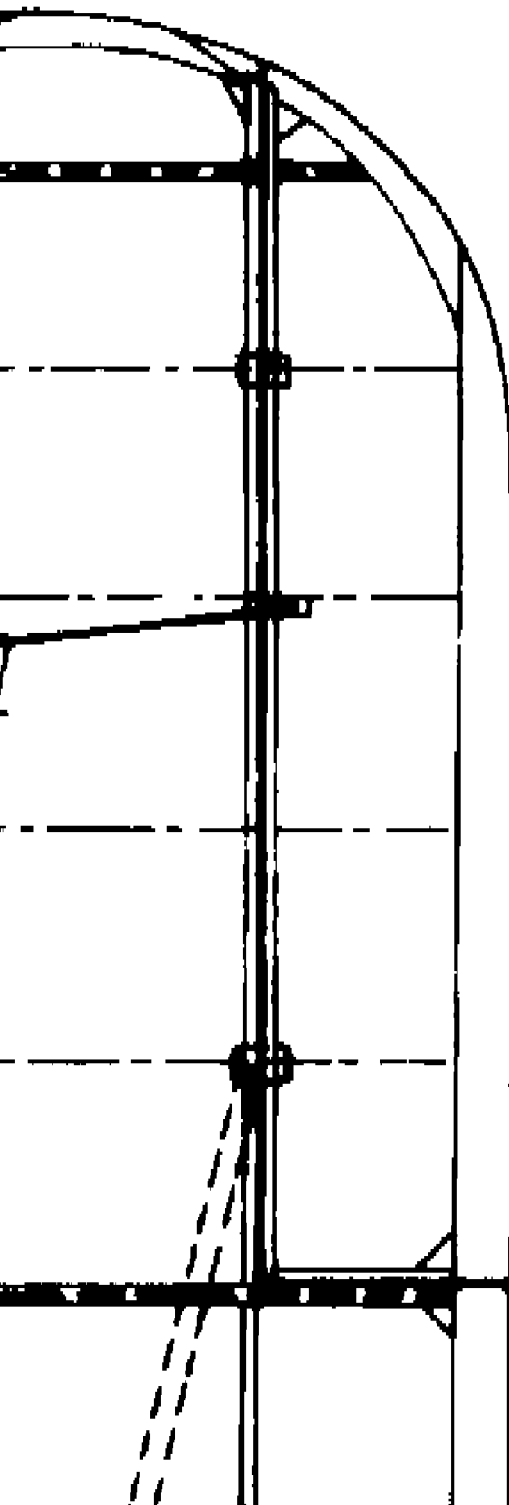


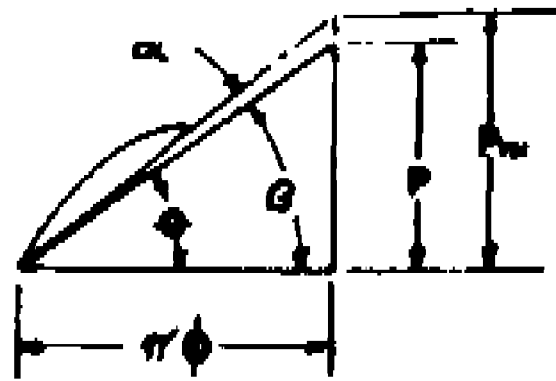
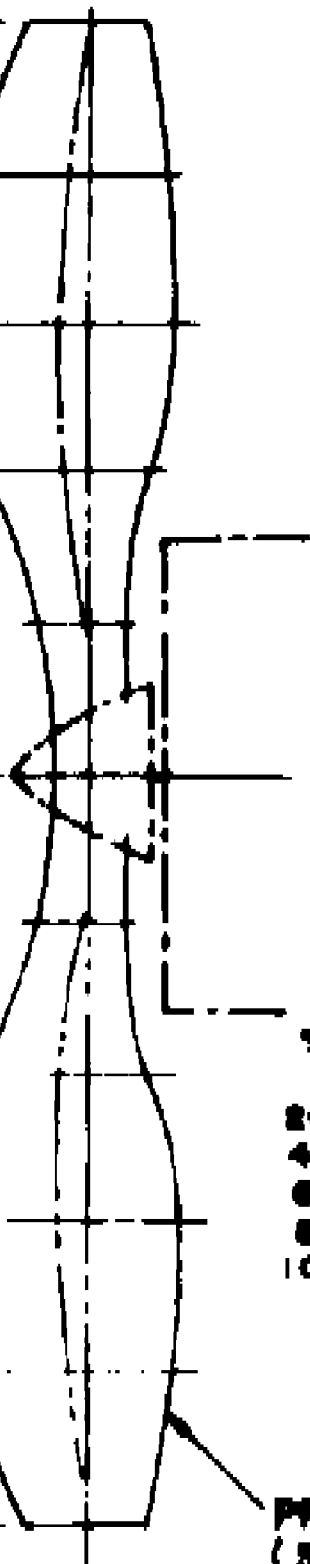
1/16" SHEET

1/16" x 5/32" CONTOURED
AFTER SHEETING

.020" SHEET
.020" RIBS
.020" x .080" CAP STRIPS
ALL RIBS







- ϕ BLADE STATION DIA.
- P PITCH, ACTUAL
- P_{TH} PITCH, THEORETICAL
- α PITCH ANGLE, ACTUAL
- β PITCH ANGLE, THEORETICAL
- CL ANGLE OF ATTACK

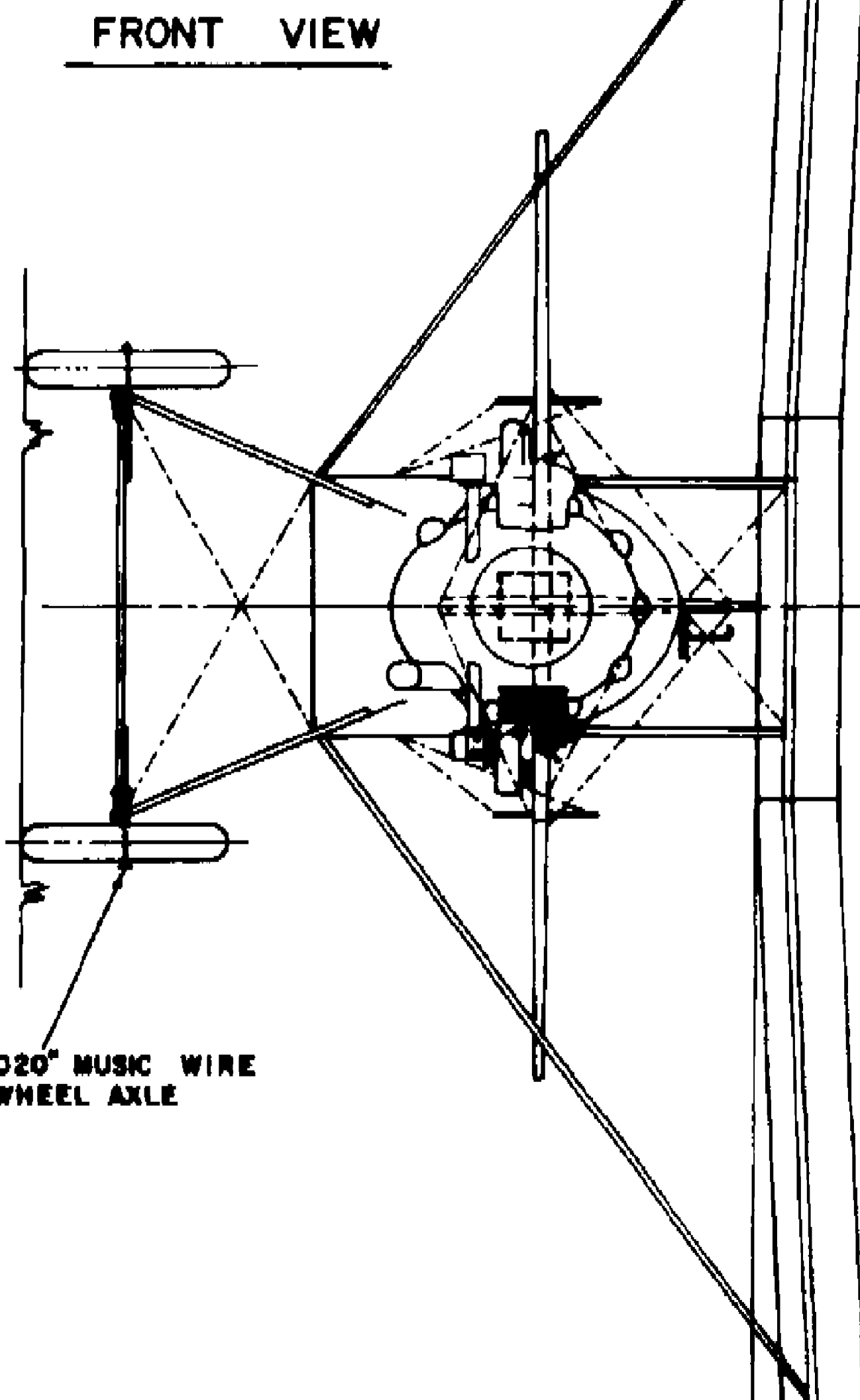
$\% \phi$	ϕ	r	P	$\frac{P}{r}$	$\alpha = \tan^{-1} \frac{P}{r}$	CL	$\beta = \alpha + CL$	P_{TH}
0	0.0	0.0	14.80		90.0°	0.0°	90.0°	14.7
20	1.80	8.027		2.864	70.88	1.0	71.88	
40	3.20	10.088		1.444	65.27	2.0	67.27	
60	4.80	15.080		.962	43.88	3.0	46.88	
80	6.40	20.108		.721	36.80	3.0	39.80	
100	8.00	25.128	14.80	.577	30.00	2.0	32.00	15.7

PROPELLER DETAILS

PROJECTED PROFILE
(SIDE VIEW OF PROP BLOCK)

.005" - .010 SOFT
ALUMINUM HINDS

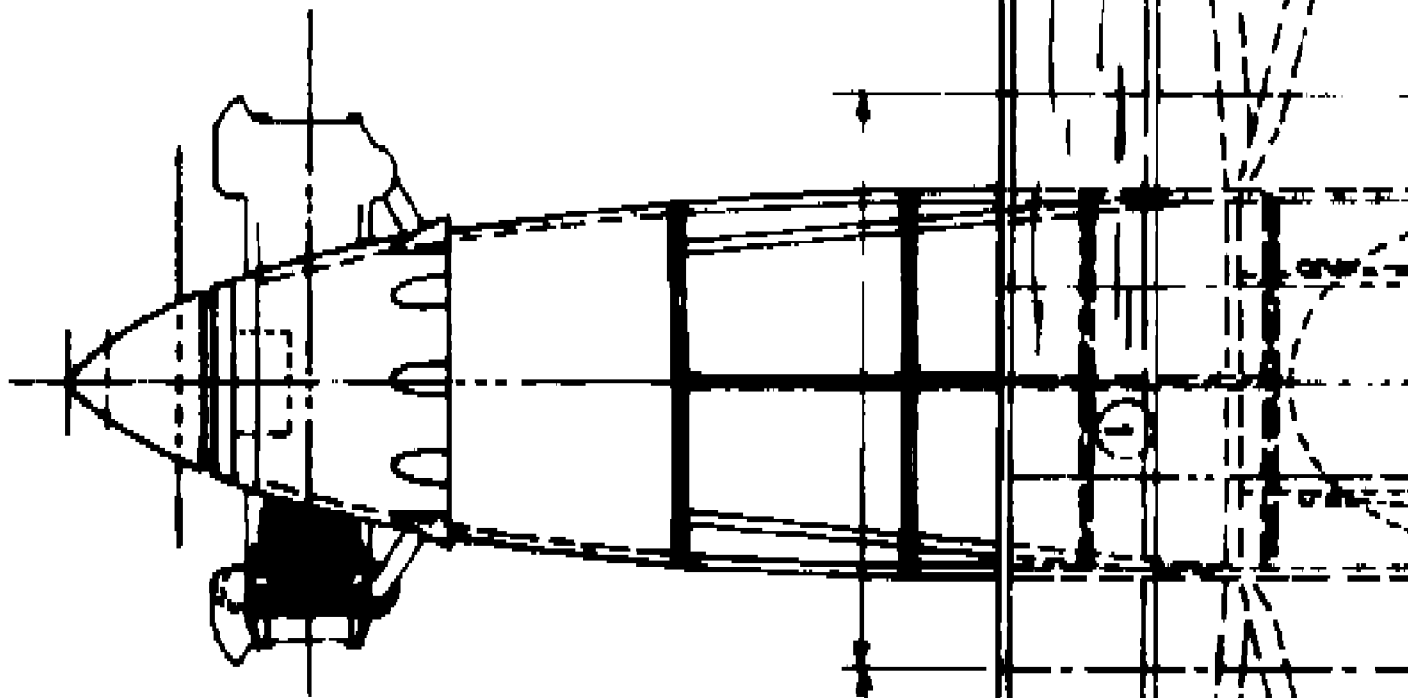
FRONT VIEW



.020" MUSIC WIRE
WHEEL AXLE

OVER
REAR
SHEE
ATT

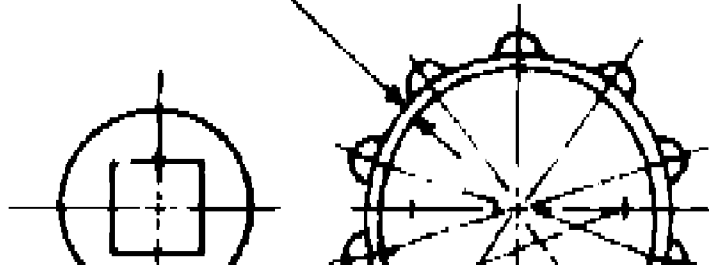
TOP VIEW



OVER THIS LENGTH FILL IN FRONT &
REAR SPARS - FULL DEPTH, WITH 1/32"
SHEET - VERTICAL GRAIN FOR STRUT
ATTACHMENTS.

1/16" x 5/32" HARD Balsa,
ALL WING STRUTS

1/16" x 3/32" WALL THICKNESS

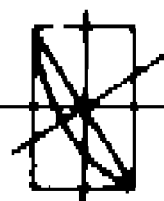


PROJECTED PLATFORM
(PLAN VIEW OF PROP. BLOCK)

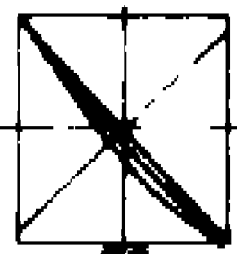
100



20% 0



40

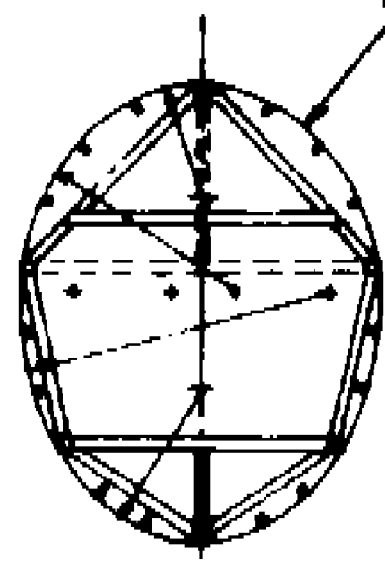


60

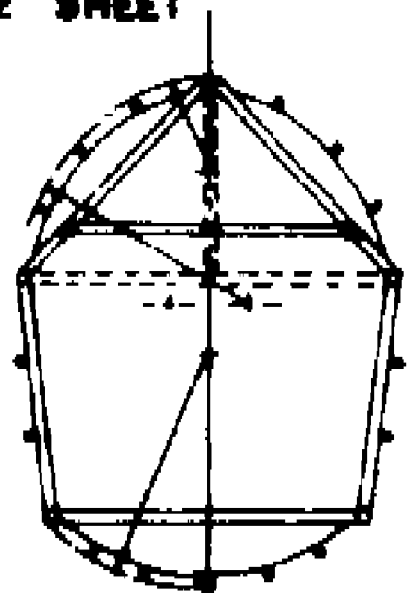
1/16" SQ

1/16" x 5/16"

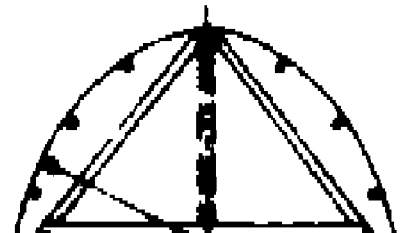
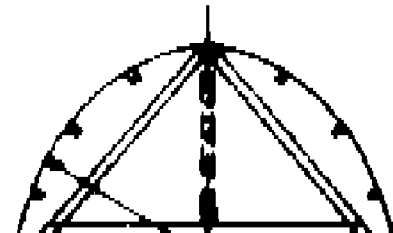
1/32" SHEET



3

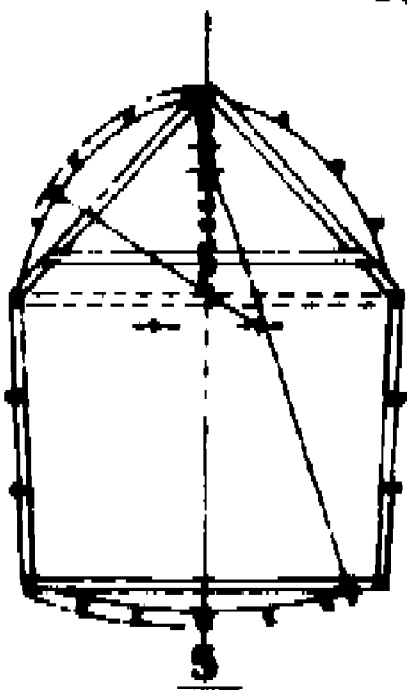
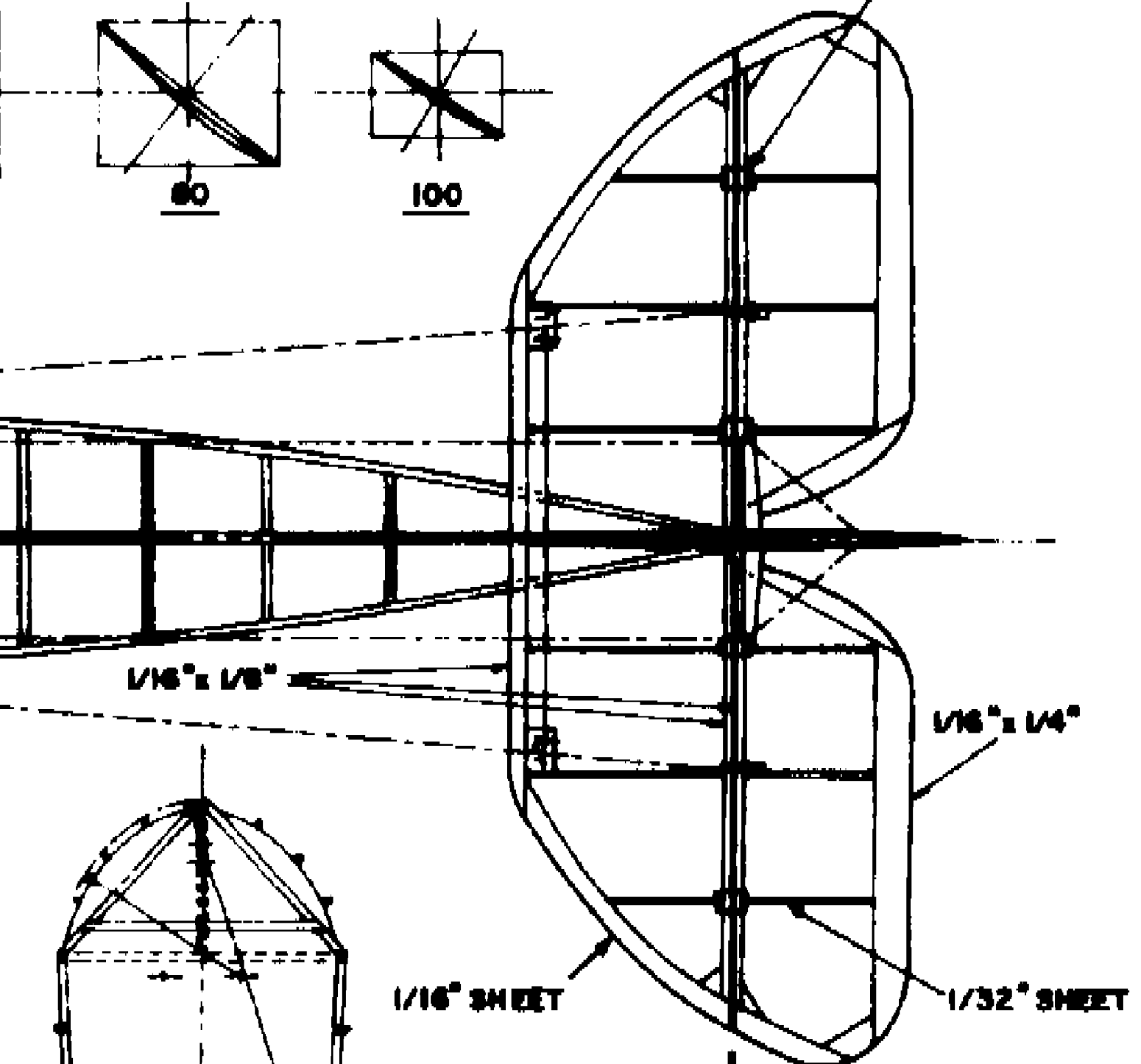
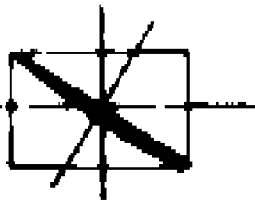
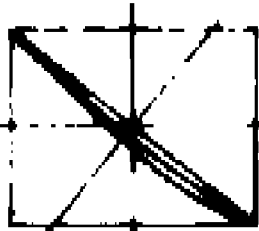


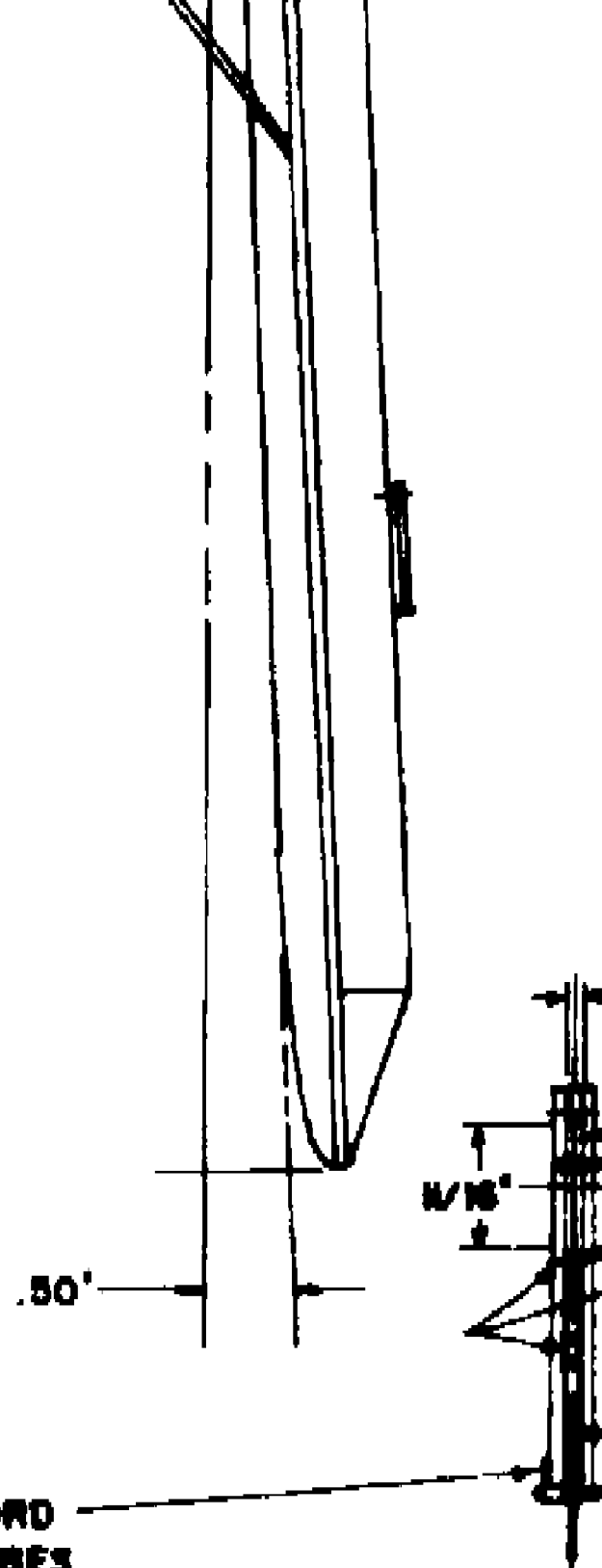
4



PROJECTED PROFILE
(SIDE VIEW OF PROP BLOCK)

.005" - .010 SOFT
ALUMINUM HINDS



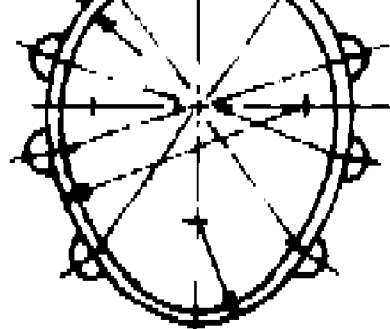
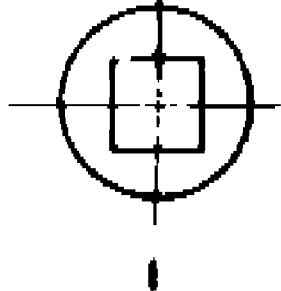


**SIMULATED SHOCK CORD
(BUNGEE) LASHING - USES
VERY THIN ELASTIC**

SECTION A

SHOWING SPLIT





2

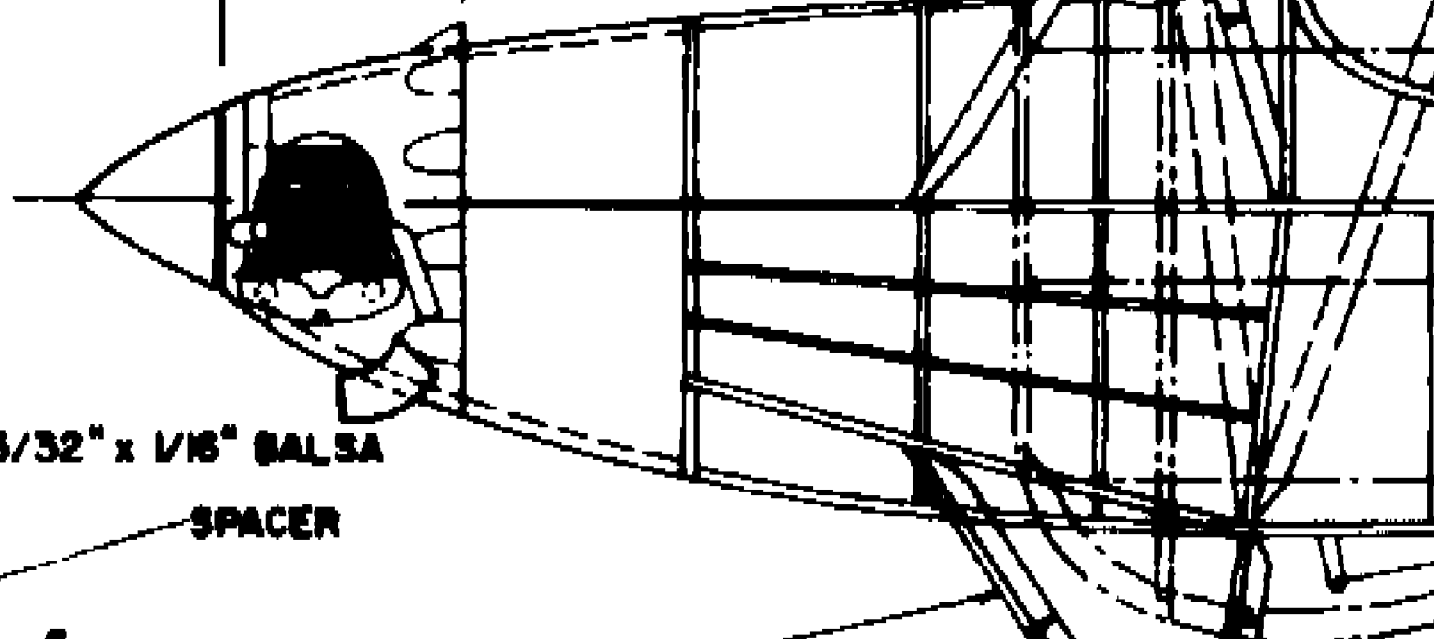
2

3

4

5

6



3/32" x 1/16" Balsa

SPACER

1/16" PIVOT .020" M.W.

SWING AXLE

1/16" DIA. BASSWOOD
DOWELS WITH SOFT
BALSA AFT. BODY

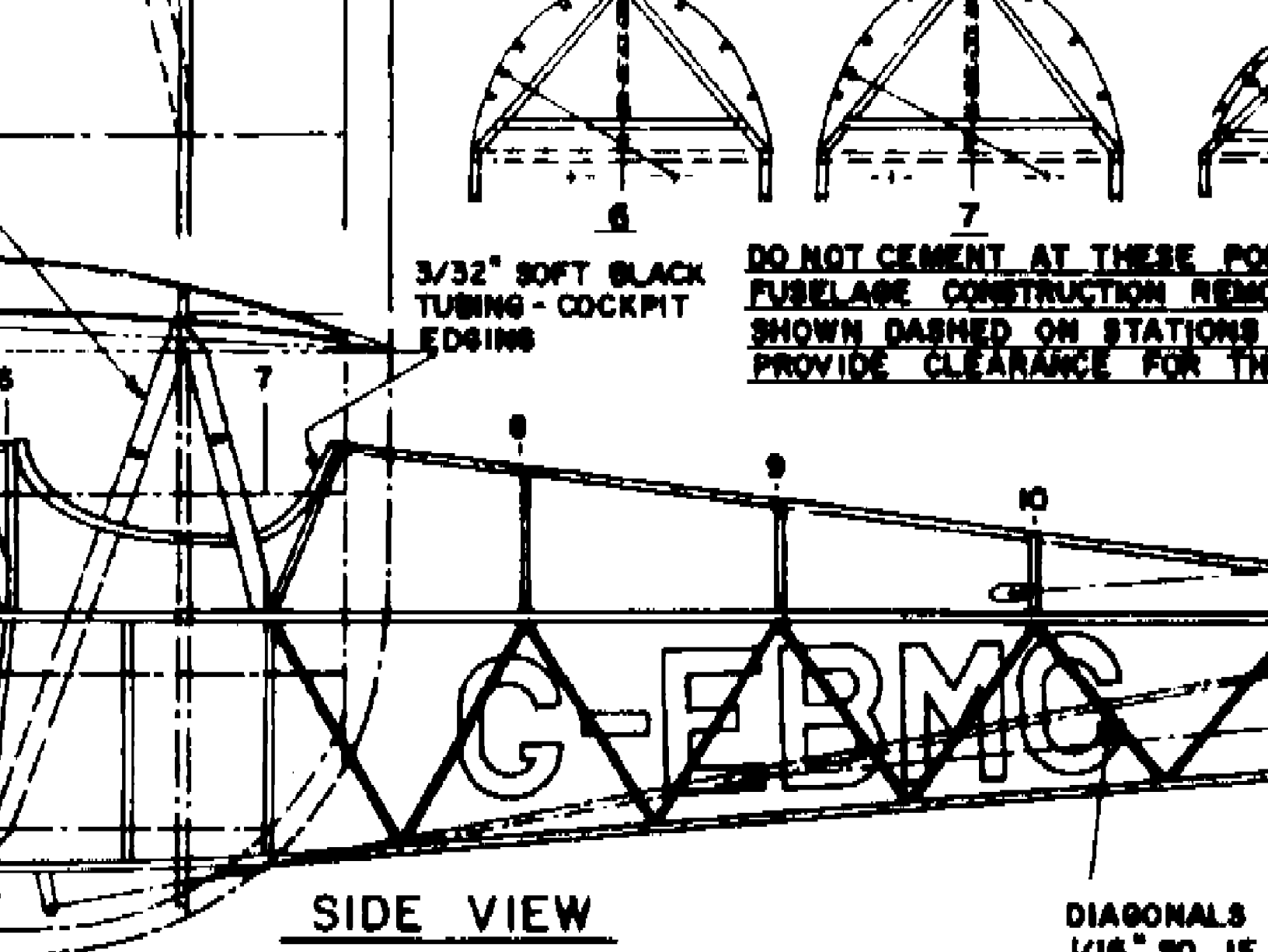
CT. A-A

A

A

SPLIT CROSS AXLE

.020" DIA. M.W.
RETAINER



SIDE VIEW

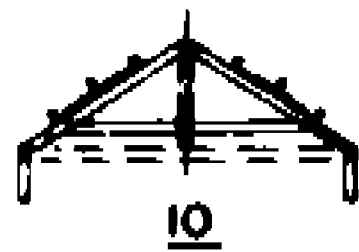
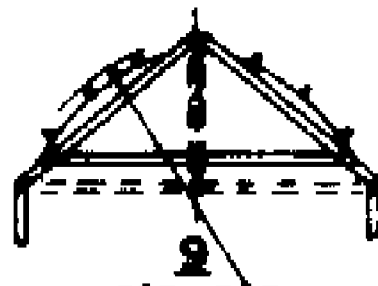
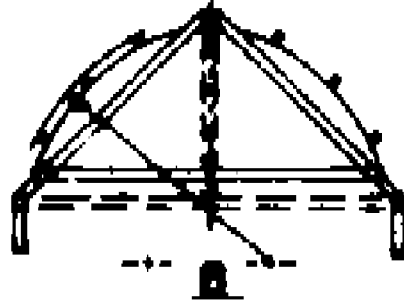
DIAGONALS
1/16" SQ. IF

LONGERONS & TRUSSES 1/16" SQ.
STRINGERS 1/32" x 1/8"

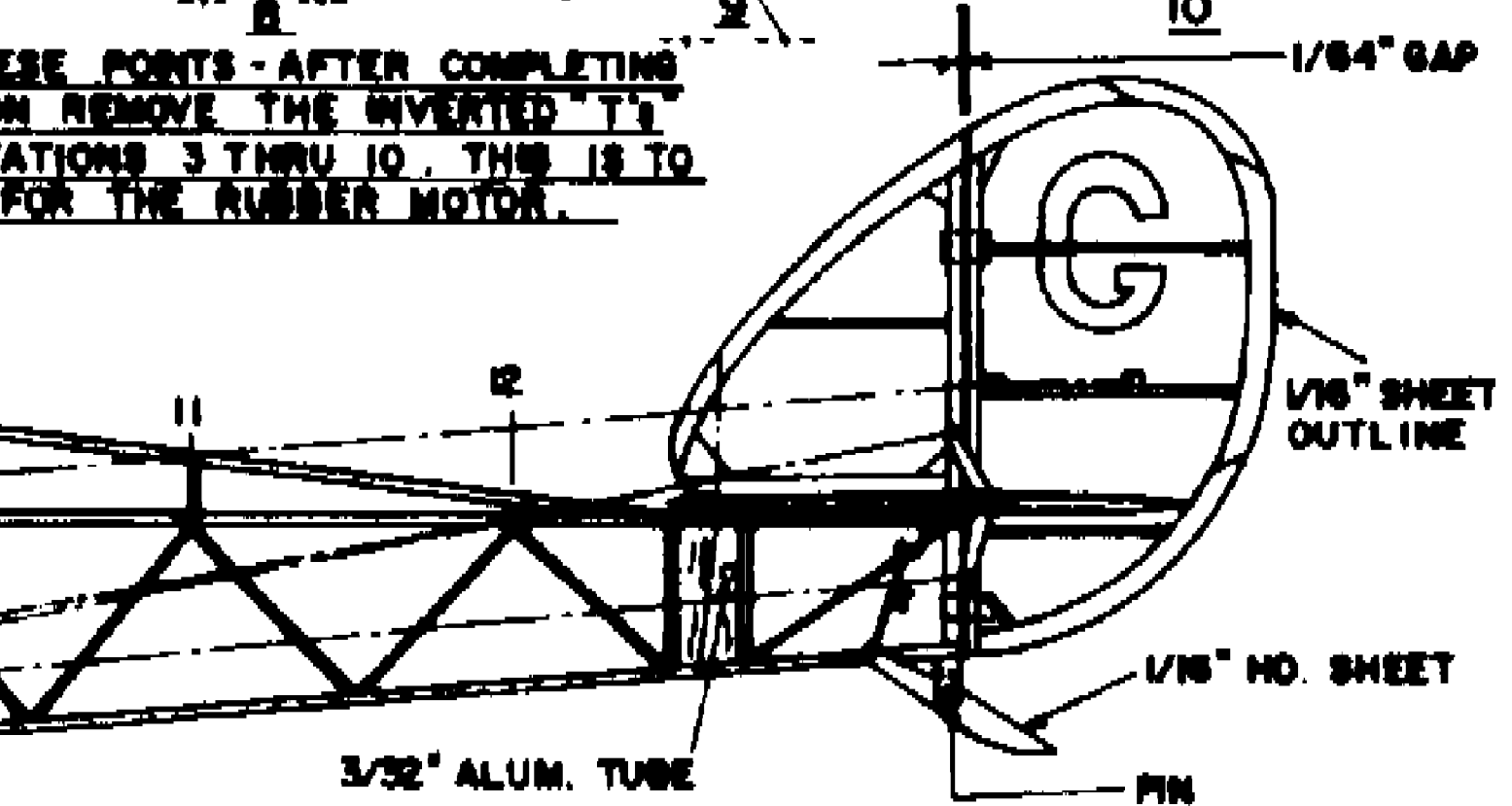
CRANWELL CLA 3 WAS DESIGNED BY FLIGHT LIEUT.
N. COMPER. BUILT BY THE CRANWELL LIGHT
AEROPLANE CLUB FOR THE 1925 LYMPHE RACES
WING SPAN = 21', ENGINE - BRISTOL CHERUB
REF: "AVIATION", SEPT. 7, 1925

JANES "ALL THE WORLD'S AIRCRAFT", 1926
"ULTRALIGHTS", BY RICHARD RIDING

SCALE 1-1/7" = 1'



REMOVE PORTS - AFTER COMPLETING
 THEN REMOVE THE INVERTED "T'S"
 OPERATIONS 3 THRU 10. THIS IS TO
 MAKE ROOM FOR THE RUBBER MOTOR.



WINGS ARE $1/32" \times 1/16"$,
 SQUARE IF DESIRED

WING LIEUT.
 WING
 RACES.
 RUB

1926

FLYING MODELS	
CRANWELL CLA 3	
MODEL DESIGNED BY J. B. GRANT	
INKED PLAN BY JOE DEMARCO	
ALL RIGHTS RESERVED	SHEET 1 OF 1